

# Effect Of Nonlinear Intensity On Orbital Stability And Chaos Control In The Generalized Camassa-Holm Equation

**Cui Su**

Faculty of Science  
Jiangsu University, Jiangsu, 212013  
Zhenjiang, China  
sucui@ujs.edu.cn

**Mengjiao Tian**

Faculty of Science  
Jiangsu University, Jiangsu, 212013  
Zhenjiang, China  
mengjiaotian@163.com

**Yanmin Wu**

Wuxi Automobile Engineering Secondary  
Professional School  
Jiangsu, Wuxi, China  
1198660249@qq.com

**Jiuli Yin\***

Faculty of Science  
Jiangsu University, Jiangsu, 212013  
Zhenjiang, China  
yjl@ujs.edu.cn

**Abstract**—Existence and stability of solitary waves in the generalized Camassa-Holm equation is considered. The nonlinear intensity has important influence on the shape and stability of solitary waves. When the power of nonlinear term is odd, the equation admits positive solitary waves which are also proved to be orbitally stable when the wave velocity exceeds a critical value. When the power of nonlinear term is even, the equation admits positive and negative solitary waves which are proved to be orbitally stable for any wave velocity. Using the Menikov method, all solitary waves turn to chaos under the external periodic perturbation with arbitrary nonlinear intensity. By applying a feedback controller, chaos can be controlled into a stable state. Results show that the uncontrollable region becomes smaller and the awful frequency appears less with the increase of nonlinear intensity.

**Keywords**—Camassa-Holm equation; Stability; Chaos; Solitary waves

## I. INTRODUCTION

The generalized Camassa-Holm equation

$$u_t - u_{xxt} + au^n u_x = 2u_x u_{xx} + uu_{xxx}, \quad (1.1)$$

is a famous shallow water equation [1]. Here  $n$  denotes nonlinear intensity and  $a(>0)$  is the parameter of nonlinear term. Eq. (1.1) has three important models and it has been also widely studied by many researchers[2-4].

When  $n=1$  and  $a=3$ , Eq. (1.1) becomes the famous Camassa-Holm equation [5-7], which has a nonsmooth solitary wave  $ce^{-|x-ct|}$  (where  $c$  is the wave speed) which is called a peakon. This special solution is proved to be orbitally stable for any wave speed [8].

Eq. (1.1) is reduced to the modified Camassa-Holm equation as  $n=2$  [9-13], which admits negative and positive smooth solitary waves. This negative smooth solution is proved to be orbitally stable for any wave speed [10]. Eq. (1.1) becomes the Camassa-Holm equation with quartic nonlinearity when  $n=3$  [14-15], which admits a positive smooth solitary wave. This positive smooth solution is proved to be orbitally stable when the speed exceeds a critical value [15]. More property of generalized Camassa-Holm equation has been extensively studied [16-17].

It is obviously that the nonlinear intensity  $n$  has a great influence on the shape and stability of the solitary wave as the above facts shown. Hence our first objective is to carry out a further study on the existence and stability of solitary waves as the nonlinear  $n > 3$  strength. Moreover,

the propagation of solitary waves is not in a pure environment, and it is easy to be affected by external perturbation. It is nature that the second objective of this paper is to study evolution progress and control problem of solitary wave under the external perturbation.

The rest of the paper is organized as follows. In Section 2, existence of solitary wave and homoclinic orbits of Eq. (1.1) are given. In Section 3, stability of solitary waves is considered. In Section 4, dynamics behavior for the perturbed and controlled system are studied.

## II. EXISTENCE OF SOLITARY WAVE

The solitary wave of Eq. (1.1) has the form as

$$u(x,t) = \phi_c(x-ct) \quad (2.1)$$

and the profile  $\phi_c$  propagating at speed  $c > 0$ .

Then Eq. (1.1) becomes

$$-c\phi_{cx} + c\phi_{cxxx} + a\phi_c^n \phi_{cx} = 2\phi_{cx}\phi_{cxx} + \phi_c\phi_{cxxx} \quad (2.2)$$

Using the decay of  $\phi_c$  at infinity, we obtain

$$c\phi_c - c\phi_{cxx} - \frac{a}{n+1}\phi_c^{n+1} + \phi_c\phi_{cxx} + \frac{\phi_{cx}^2}{2} = 0 \quad (2.3)$$

Multiplying by  $\phi_{cx}$  and integrating Eq. (2.3), we have

$$\phi_{cx}^2 = \frac{\phi_c^2(c - \frac{a\phi_c^n}{2(n+2)})}{c - \phi_c} \quad (2.4)$$

Remark 1. We can deduce that solitary waves from Eq. (2.4) exist for  $c > \sqrt[n-1]{\frac{2(n+2)}{a}}$ .

Theorem 1. For any  $n \in \mathbb{Z}^+$ , Eq. (1.1) admits homoclinic orbits associated with the solitary waves as  $a > 0$ .

Proof. Letting  $y = \phi'$ , Eq. (2.3) becomes

$$\begin{cases} \phi' = y, \\ y' = \frac{-c\phi + \frac{a}{n+1} - \frac{y^2}{2}}{\phi - c} \end{cases} \quad (2.5)$$

Taking the commutation  $d\xi = (\phi - c)d\tau$ , Eq. (2.5) is rewritten as

$$\begin{cases} \frac{d\phi}{d\tau} = (\phi - c)y, \\ \frac{dy}{d\tau} = -c\phi + \frac{a}{n+1}\phi^{n+1} - \frac{y^2}{2} \end{cases} \quad (2.6)$$

Hence Eq. (2.6) has the Hamiltonian function as follows

$$H(\phi, y) = (c - \phi)y^2 + \frac{a}{2(n+2)}\phi^{n+2} - c\phi^2 \quad (2.7)$$

Case I. When  $n$  is even number, Eq. (2.6) has three equilibrium points: A(0,0), B  $(-\sqrt[n]{\frac{(n+1)c}{a}}, 0)$ , C  $(\sqrt[n]{\frac{(n+1)c}{a}}, 0)$ .

Case II. When  $n$  is odd, Eq.(2.6) has two equilibrium points: D(0,0), E  $(\sqrt[n]{\frac{(n+1)c}{a}}, 0)$ .

According to the dynamics theory, we know the system (2.6) admits homoclinic orbits (see Fig. 1(a), Fig. 2(c)). Furthermore, system (2.6) also has the solitary waves (see Fig. 1(b), Fig. 2(d)).

We find that Eq. (1.1) only admits positive solitary waves when  $n$  is odd, while both positive and negative solitary waves appear when  $n$  is even. It is noted that

the height of the solitary wave is smaller and tends to

$$\sqrt[n]{\frac{(n+1)c}{a}}$$

with the increasing  $n$ .

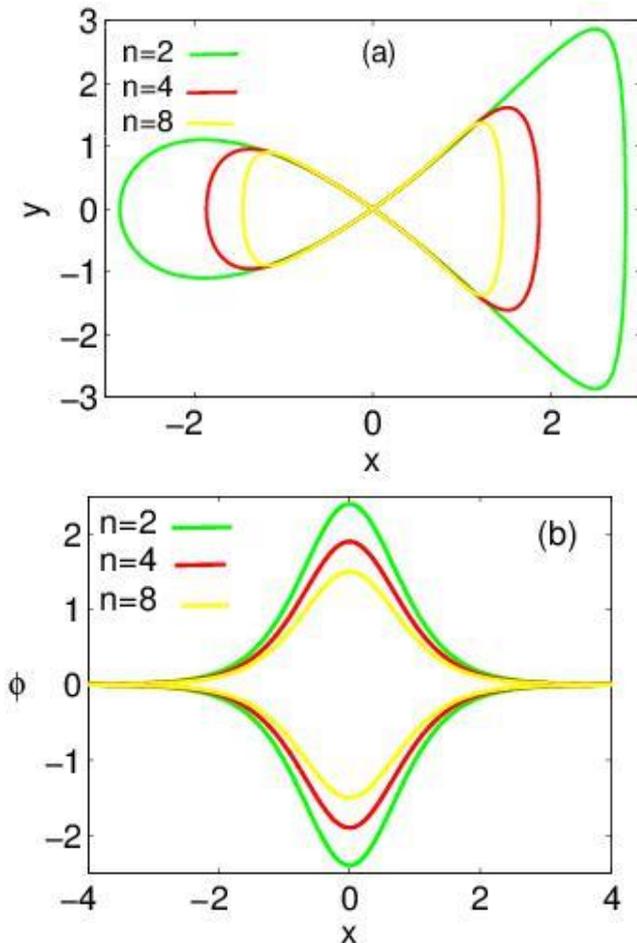


Fig. 1. (a) phase portrait of Eq. (2.7); (b) The negative and positive solitary wave.

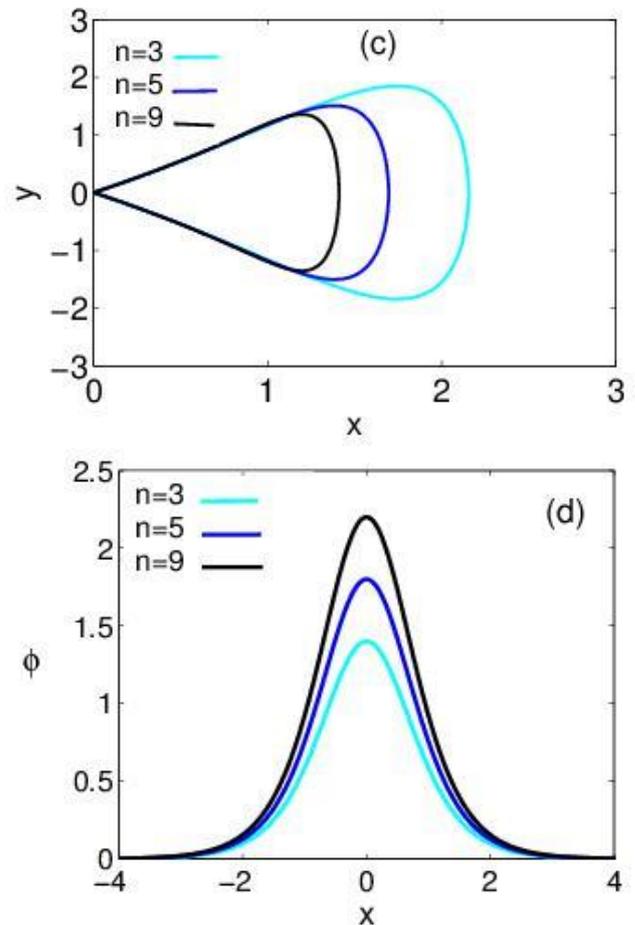


Fig. 2. (c) phase portrait of Eq. (2.7); (d) The positive solitary wave.

### III. STABILITY OF SOLITARY WAVE

Eq. (1.1) has two invariants as follows

$$\begin{cases} E(u) = \frac{1}{2} \int_R (u^2 + u_x^2) dx, \\ F(u) = -\frac{1}{2} \int_R \frac{au^{n+2}}{2(n+2)} + uu_x^2 dx. \end{cases} \quad (3.1)$$

Eq. (2.3) can be also described as

$$F'(\phi_c) + cE'(\phi_c) = 0, \quad (3.2)$$

Where represents the Frechet derivatives of  $E$  and  $F$  and the linearized operator is given as

$$H_c = F''(\phi_c) + cE''(\phi_c) = -\partial_x (c - \phi_c) \partial_x - a\phi_c^n + \phi_{cxx} + c.$$

It is noted that the functions  $\phi_c, \phi_{cx}, \phi_{cxx} \rightarrow 0$

exponentially fast when  $|x| \rightarrow \infty$ . The Liouville transformation admits the following expression

$$\psi(y) = (2c - 2\phi_c(z))^{-\frac{1}{n+1}} v(x),$$

$$y = \int_0^x \frac{1}{\sqrt{c - \phi_c(z)}} dz.$$

Taking advantage of spectral equation  $H_c v = \lambda v$  and Liouville transformation, we have

$$F_c \psi(y) = (-\partial_y^n + p_c(y) + c)\psi(y) = \lambda \psi(y),$$

where  $p_c(y) = -a\phi_c^n(x) + \frac{3}{4}\phi_c'' - \frac{(\phi_c'(x))^2}{8(c - \phi_c(x))}$ .

The stability depends on the convexity properties of the function  $d(c) = F(\phi_c) + cE(\phi_c)$ . We have the following lemma.

Lemma 1. According to the literature, the stability we know that the solitary wave  $\phi_c$  is unstable if  $d''(c) < 0$  and stable if  $d''(c) > 0$ .

By differentiating  $d(c)$  once, we get

$$d'(c) = \left\langle F'(\phi_c) + cE'(\phi_c), \frac{\partial \phi_c}{\partial c} \right\rangle + E(\phi_c) = E(\phi_c). \tag{3.3}$$

Case 1. Positive solitary wave

In this case, the solitary wave  $\phi_c \in [0, \sqrt{\frac{(n+1)c}{a}}]$

and  $\phi_c$  is an even function. So we have

$$d''(c) = \frac{d}{dc} \int_R \frac{1}{2} (\phi_c^2 + \phi_{cx}^2) dx$$

$$= \frac{d}{dc} \int_0^{+\infty} \phi_c^2 \frac{2c - \phi_c - \frac{a\phi_c^n}{2(n+2)}}{c - \phi_c} dx$$

$$= \frac{d}{dc} \int_0^{+\infty} \phi_c(-\phi_{cx}) \frac{\sqrt{\frac{c - \phi_c}{c - \frac{a\phi_c^n}{2(n+2)}}} \cdot \frac{2c - \phi_c - \frac{a\phi_c^n}{2(n+2)}}{c - \phi_c}}{\sqrt{\frac{c - \phi_c}{c - \frac{a\phi_c^n}{2(n+2)}}}} dx$$

$$= \frac{d}{dc} \int_0^{+\infty} \phi_c(-\phi_{cx}) \frac{4(n+2)c - 2(n+2)\phi_c - a\phi_c^n}{\sqrt{2(n+2)}\sqrt{2(n+2)c - a\phi_c^n}\sqrt{c - \phi_c}} dx$$

$$= \frac{d}{dc} \int_0^{\sqrt{\frac{2(n+2)c}{a}}} y \frac{-4(n+2)c + 2(n+2)y + ay^n}{\sqrt{2(n+2)}\sqrt{2(n+2) - ay^n}\sqrt{c - y}} dy.$$

We can get a simpler form if letting  $y = \sqrt{\frac{2(n+2)c}{a}} s$

as follows

$$d''(c) = \frac{d}{dc} \int_0^1 \frac{(\frac{2(n+2)c}{a})^{\frac{2}{n}} s [4(n+2)c - 2(n+2)\sqrt{\frac{2(n+2)c}{a}} s - 2(n+2)cs^n]}{\sqrt{2(n+2)}\sqrt{2(n+2)c - 2(n+2)cs^n}\sqrt{c - (\frac{2(n+2)c}{a})^{\frac{1}{n}} s}} ds$$

$$= \frac{d}{dc} \int_0^1 \frac{(\frac{2(n+2)c}{a})^{\frac{2}{n}} s (2c - \sqrt{\frac{2(n+2)c}{a}} s - cs^n)}{\sqrt{c - cs^n}\sqrt{c - (\frac{2(n+2)c}{a})^{\frac{1}{n}} s}} ds. \tag{3.4}$$

Case 2. Negative solitary wave

In this case, the solitary wave  $\phi_c \in [-\sqrt{\frac{(n+1)c}{a}}, 0]$

and  $\phi_c$  is an even function. So we have the following fact

$$d''(c) = \frac{d}{dc} \int_R \frac{1}{2} (\phi_c^2 + \phi_{cx}^2) dx$$

$$= \frac{d}{dc} \int_0^{+\infty} \phi_c^2 \frac{2c - \phi_c - \frac{a\phi_c^n}{2(n+2)}}{c - \phi_c} dx$$

$$= \frac{d}{dc} \int_0^{+\infty} \phi_c(-\phi_{cx}) \frac{4(n+2)c - 2(n+2)\phi_c - a\phi_c^n}{\sqrt{2(n+2)}\sqrt{2(n+2)c - a\phi_c^n}\sqrt{c - \phi_c}} dx$$

We can also get a simpler form if letting

$$y = -\sqrt[n]{\frac{2(n+2)c}{a}}s \text{ as follows}$$

$$d''(c) = \frac{d}{dc} \int_0^1 \frac{(\frac{2(n+2)c}{a})^{\frac{2}{n}}s(2c + \sqrt[n]{\frac{2(n+2)c}{a}}s - cs^n)}{\sqrt{c - cs^n}\sqrt{c + (\frac{2(n+2)c}{a})^{\frac{1}{n}}s}} ds. \tag{3.5}$$

(3.5)

It is hard to get the integrable value according to the Eq.(3.4) or Eq. (3.5). So we will use MATLAB to compute the value of  $d''(c)$ .

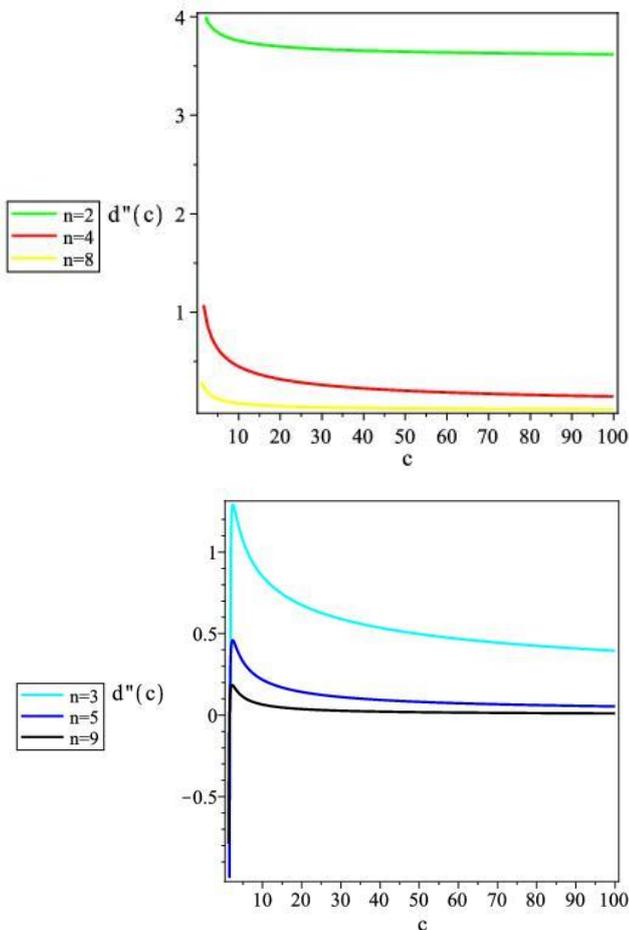


Fig. 3. (a) The integration value of Eq.(3.4); (b) The integration value of Eq.(3.5).

Remark. Fig. 3(a) shows that the positive solitary wave is unstable when the speed  $c$  is close to the critical value  $\sqrt[n-1]{\frac{2(n+2)}{a}}$ , which connects with the bifurcation condition contributes. And the positive solitary wave is stable when the speed is slightly greater than the critical value. Meanwhile, Fig.3(b) shows that the negative solitary wave is stable for any wave speed.

#### IV. DYNAMICS BEHAVIOR FOR THE PERTURBED AND CONTROLLED SYSTEM

The perturbed equation (2.7) is given as:

$$\begin{aligned} \frac{d\phi}{d\tau} &= (c - \phi)y, \\ \frac{dy}{d\tau} &= -c\phi + \frac{a}{n+1}\phi^{n+1} - \frac{y^2}{2} + r \cos \omega\tau, \end{aligned} \tag{4.1}$$

where  $r$  and  $\omega$  denote the amplitude and the frequency of perturbation term respectively. The unperturbed system permits a homoclinic orbit given by Eq. (4.1). It is noted that the closed homoclinic orbits perhaps break under the perturbation. So we want to find the criteria of the existence of homoclinic bifurcation and chaos by the melnikov method.

Supposing the unperturbed homoclinic orbits written as  $(x_0, y_0) = (x^\pm(\tau), y^\pm(\tau))$ , the Melnikov function for system (4.1) can be given by

$$\begin{aligned} M(\tau_0) &= \int_{-\infty}^{+\infty} y_0(\tau)r \cos \omega(\tau + \tau_0)d\tau \\ &= -2r \sin \omega\tau_0 \int_0^{+\infty} y_0(\tau) \sin \omega\tau d\tau \\ &= -2rI \sin \omega\tau_0, \end{aligned}$$

Where  $I = \int_0^{\infty} y_0(\tau) \sin \omega\tau d\tau$  is a function of frequency  $\omega$ .

By using the Melnikov's theorem [18], the chaos occurs if  $M(\tau_0) = 0$  and  $M'(\tau_0) \neq 0$  for some  $\tau_0$ . It is easy to find that the Melnikov function has the value  $\tau_0 = 0$  such that  $M(0) = 0$  and  $\frac{\partial M}{\partial \tau_0} \Big|_{\tau_0=0} = -2rI \neq 0$ . Hence we can deduce that chaos would occur under the one external perturbations.

In order to suppressing chaos, the controlled system of Eq. (4.1) is devised as

$$\begin{aligned} \frac{d\phi}{d\tau} &= (c - \phi)y, \\ \frac{dy}{d\tau} &= -c\phi + \frac{a}{n+1}\phi^{n+1} - \frac{y^2}{2} + r \cos \omega\tau - ky. \end{aligned} \quad (4.2)$$

And the relative Melnikov function is yielded as

$$M(\tau_0) = -k \int_{-\infty}^{+\infty} y_0^2(\tau) d\tau + \int_{-\infty}^{+\infty} y_0(\tau) r \cos \omega(\tau + \tau_0) d\tau.$$

Considering  $y_0(\tau)$  is odd, we obtain

$$\begin{aligned} M(\tau_0) &= -2k \int_0^{+\infty} y_0^2(\tau) d\tau - 2r \sin \omega\tau_0 \int_0^{+\infty} y_0(\tau) \sin \omega\tau d\tau \\ &= -2kC_{h0} - 2rI_1 \sin \omega\tau_0, \end{aligned}$$

where  $C_{h0} = \int_0^{+\infty} y_0^2(\tau) d\tau$  and

$$I_1 = \int_0^{+\infty} y_0(\tau) \sin \omega\tau d\tau.$$

Based the analysis in Ref. [15], the chaos may be controlled as  $k \geq \frac{|rI_1(\omega)|}{C_{h0}} = R(\omega)$ . In order to investigate the influence of different parameters on control ability, some numerical results are given in Fig. 4 and Fig. 5. The graph of  $k = \frac{|rI_1(\omega)|}{C_{h0}}$  shows the following facts:

- (1) The uncontrollable region becomes larger with the increase of perturbation amplitude  $r$ .
- (2) The awful frequency appears periodicity where chaos is extremely difficult to be controlled.

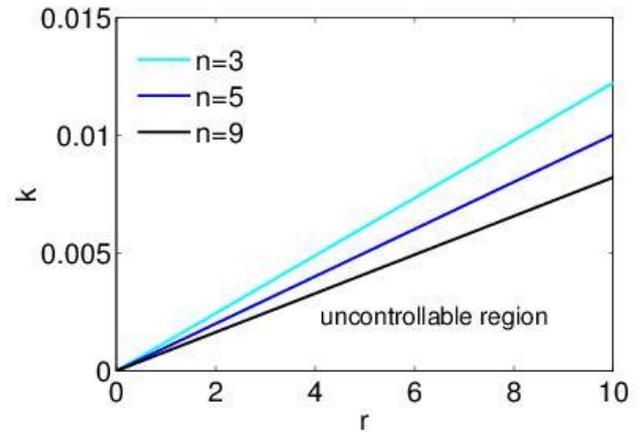


Fig. 4. Chaos threshold for the perturbed system with  $a = 3, c = 4, \omega = 0.5$ .

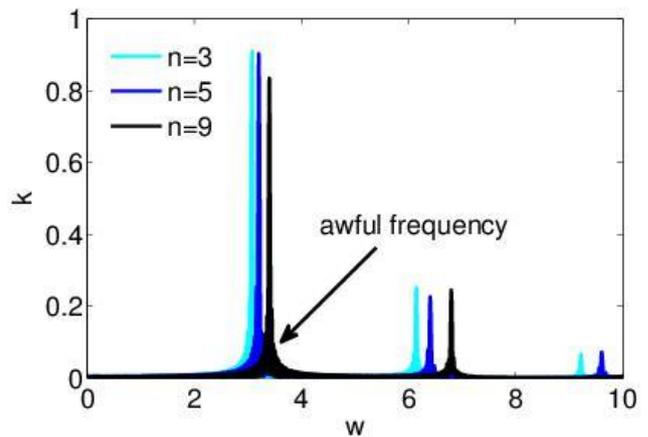


Fig. 5. Chaos threshold for the perturbed system with  $a = 3, c = 4, r = 1.2$ .

- (3) The uncontrollable region becomes smaller and the awful frequency appears less with the increase of nonlinear intensity  $n$ .

## V. CONCLUSION

Based on the generalized Camassa-Holm equation, we study the existence and stability of solitary waves and find the nonlinear intensity has important influence on the shape and stability of solitary waves. Following conclusions could be obtained,

- (1) When the power of nonlinear term is odd, the equation admits positive solitary waves which are also

proved to be orbitally stable when the wave velocity exceeds a critical value.

(2) When the power of nonlinear term is even, the equation admits positive and negative solitary waves which are proved to be orbitally stable for any wave velocity.

(3) All solitary waves turn to chaos under the external periodic perturbation with arbitrary nonlinear intensity.

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